Amendments to the Specification:

Please replace the first 2 paragraphs of the Detailed Description on page 3, lines 13-31 with the following amended paragraphs:

Referring to the Figures, which show an example of one embodiment of the invention, Figure 1 shows a connector with a first store-side part 10 and a second aircraft-side part 12. (Store-side part 10 may be referred to throughout this document as first part, first half, or first connector half. Aircraft-side part 12 may be referred to throughout this document as second part, second half, or second connector half.) The second aircraft-side part comprises a core 14 containing the required electrical conductors and contacts 56 at a forward end 52 near an electric cable 54. The electrical contacts 56 are surrounded by a barrel 16 having keys 18 for rotational alignment of the first and second parts 10, 12 and proper registration of the electrical contacts 56 in them. The aircraft-side part 12 further comprises an attachment ring 20 to which ends of a lanyard 22 are anchored, and an outer shell 50 assembled from a molded sleeve 28, an internally threaded clamping ring 30 and resilient metallic (e.g., spring steel) fingers 24 integrally formed with a mounting ring 26.

To assemble the aircraft-side part 12, as shown in Figure 2, the attachment ring 20 is slid over the barrel up to a shoulder 38 on the core 14. The attachment ring 20 is shown with a lanyard 22 fitted. However, alternate embodiments could include an alternative cable strain relief device or interface. The next component fitted over the barrel 16 is mounting ring 26 and fingers 24, which are pre-assembled on the sleeve 28 together with an electro-magnetic compatibility (EMC) shielding ring 32. This assembly 24, 26, 28 is retained over the barrel by the clamping ring 30, which is screwed onto eternal threads on the core 14. Tightening the ring 30 clamps the sleeve 28 and mounting ring 28 against a further shoulder 40 formed on the core 14. The attachment ring 20 is thereby trapped for free rotation on the core 14 between the shoulder 38 and the mounting ring 26.